NSN 2840-00-113-4074

Aircraft Gas Turbine Engine Turbine Shroud Segment - Page 1 of 1



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End Application:
Aircraft eng tf-39
Design Type:
Compressor
Blade Attachment Method:
Pin
Criticality Code Justification:
Agav
Engine Section Location:
Turbine
Stage Number:
2
Special Features:
T.O. 2j-tf39-4; goes on the fan stator assembly
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli
Fiig:
A10700