NSN 2840-00-435-6872

Aircraft Gas Turbine Engine Combustion Chamber Case - Page 1 of 1



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Overall Length:
Between 16.118 inches and 16.132 inches
Diameter:
Between 41.770 inches and 41.790 inches
End Application:
Tf-33 aircraft engine
Hardness Rating:
50.0 rockwell c and 58.0 rockwell c
Criticality Code Justification:
Feat
Engine Section Location:
Combustion chamber
Special Features:
Form, fit and function
Material:
Nickel alloy
Surface Treatment:
Nickel and cadmium
Surface Treatment Specification:
Ams2416 assn standard single treatment response
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli
Fiig:
A10700