NSN 2840-00-883-5965

Aircraft Gas Turbine Engine Turbine Rotor Blade Lock - Page 1 of 1



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Overall Length:
Between 1.315 inches and 1.365 inches
Overall Width:
Between 0.264 inches and 0.294 inches
End Application:
Tf-33 aircraft engine
Hardness Rating:
Between 63.0 rockwell a and 68.0 rockwell a
Stage Number:
8
Special Features:
Goes on the compressor rotor-front
Material:
Stainless steel
Material Specification:
Ams5517 assn standard single material response
Reference Data And Literature:
2j-tf33-4
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli
Filg:
A503q0