NSN 2840-01-033-9081

Aircraft Gas Turbine Engine Compressor Hub - Page 1 of 1



View Online at https://aerobasegroup.com/nsn/2840-01-033-9081

| Overall Length: |
|--|
| Between 4.004 inches and 4.014 inches |
| Diameter: |
| Between 7.162 inches and 7.166 inches |
| End Application: |
| =100-pw-100 aircraft engine |
| Design Type: |
| Compressor |
| Criticality Code Justification: |
| Feat Control of the C |
| Product Name: |
| Orig name nut, plain, round |
| Engine Section Location: |
| Compressor |
| Stage Number: |
| 2 |
| Special Features: |
| ntl involute spline data: teeth 140; pressure angle 30 deg.; t.O. 2j-f100-4, figure 115, index 3; used on the rear compressor drive; see |
| cto 2j-f10 (ii) -509 supersedes pn 4036639; f100 fracture critical part. |
| Material: |
| Nickel alloy hub |
| Material Specification: |
| Pwa1022 mfr ref single material response hub |
| Shelf Life: |
| N/a |
| Unit Of Measure: |
| |
| Demilitarization: |
| Yes - demil/mli |
| Fiig: |
| A10700 |
| |
| |
| |