NSN 2840-01-426-4379

Aircraft Gas Turbine Engine Combustion Chamber Case - Page 1 of 1



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End Application:
F110-100 aircraft gas turbine engine
Design Type:
Axial compressor
Criticality Code Justification:
Czbj and feat
Engine Section Location:
Combustion chamber
Special Features:
T.O. 2j-f110-4, fig 65; combustion diffuser nozzle
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli

Fiig: A10700