NSN 2840-01-430-2158

Aircraft Gas Turbine Engine Compressor Rotor - Page 1 of 1



View Online at https://aerobasegroup.com/nsn/2840-01-430-2158 **Overall Length:** 15.500 inches Diameter: 18.783 inches **End Application:** F100pw229 aircraft gas turbine engine **Criticality Code Justification:** Feat and czbj and abmz and abhp **Engine Section Location:** Compressor Stage Number: **Special Features:** Rotor consists of 3 stgs; disk & hub; 1stg o/a diameter on/c 14.590; o/a length 9.000; 2nd stg o/a diameter on/c 16.188; o/a length 7.970; involute spline data: no of teeth 79; spline pitch 20/40; pressure angle 30 deg.; pitch dia. 3.9500; 3rd stg o/a length 4.000; all stages recieve dovetail blades Material: Titanium alloy **Material Specification:** Pwa 1215 mfr ref single material response Shelf Life: N/a **Unit Of Measure: Demilitarization:**

Yes - demil/mli

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