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End Application:

F110-ge-129 aircraft gas turbine engine

Design Type:

Axial compressor

Engine Section Location:

High pressure compressor

Special Features:

Used on the compressor front; reworked and reidentified from p/n 9547m15g05

Reference Data And Literature:

T.O. 2j-f110-14, fig 1, ind 9; fig 55, ind 51; figures 53, 57 through 64

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

Yes - demil/mli

Fiig:

A10700