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End Application:

F110 aircraft gas turbine engine

Design Type:

Axial compressor

Blade Attachment Method:

Pin

Criticality Code Justification:

Czbj and bglw and feat

Engine Section Location:

High pressure compressor

Special Features:

T.O. 2j-f110-14, fig 62 ind 5; inlet guide vane near 1st vane, compressor

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

Yes - demil/mli

Fig:

A10700