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**End Application:**

F110 aircraft gas turbine engine

**Design Type:**

Axial compressor

**Blade Attachment Method:**

Pin

**Criticality Code Justification:**

Czbj and bglw and feat

**Engine Section Location:**

High pressure compressor

**Special Features:**

T.O. 2j-f110-14, fig 62 ind 5; inlet guide vane near 1st vane, compressor

**Shelf Life:**

N/a

**Unit Of Measure:**

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**Demilitarization:**

Yes - demil/mli

**Fig:**

A10700