## NSN 2840-01-436-9778

Aircraft Gas Turbine Engine Compressor Vane - Page 1 of 1



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Overall Length:
Between 12.861 inches and 12.901 inches
Overall Width:
2.075 inches
End Application:
F100-220 aircraft gas turbine engine
Design Type:
Axial compressor
Blade Attachment Method:
Lock
Hardness Rating:
36.0 rockwell c
Criticality Code Justification:
Czbj and abhp and abmk and feat
Engine Section Location:
Compressor
Special Features:
T.O. 2j-f100-24, fig 121, ind 8; used on the fan inlet case, tt2 sensing immersion thermocouple; use only on p/n 4082495
Material:
Steel
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli
Fiig:
A10700