NSN 2840-01-440-1697

Aircraft Turbine Engine Turbine Blade Damper - Page 1 of 1



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Overall Length:
Between 0.565 inches and 0.571 inches
Diameter:
16.788 inches
End Application:
=100-pw-229 aircraft gas turbine engine
Design Type:
Axial compressor
Hardness Rating:
34.0 rockwell c and 44.0 rockwell c
Criticality Code Justification:
Feat and czbj
Engine Section Location:
Compressor
Special Features:
T.O. 2j-f100-54, fig 99, ind 21; repl p/n 4074864 once tcto has been accomplished; retaining plate damper is symmetrical in shape and has
6 retention tabs forward and aft to preclude possible reverse installation and resulting cut-off of cooling air to the 2nd stage blades
Material:
Nickel alloy
Material Specification:
Ams5544 assn standard single material response
Shelf Life:
N/a
Unit Of Measure:
-
Demilitarization:
Yes - demil/mli
Filg:
410700