## NSN 2840-01-444-6530

Aircraft Turbine Engine Turbine Blade Damper - Page 1 of 1



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End Application:
Aircraft eng tf39
Design Type:
Axial compressor
Criticality Code Justification:
Agav
Engine Section Location:
High pressure turbine
Stage Number:
2
Special Features:
Original name damper assembly, high pressure turbine rotor-stage 2. Alsopart number g05 added per cid 219639
Shelf Life:
N/a
Unit Of Measure:
Demilitarization:
Yes - demil/mli
Fiig:
A10700