NSN 2840-01-444-7012

Aircraft Gas Turbine Engine Turbine Rotor Blade - Page 1 of 1



View Online at https://aerobasegroup.com/nsn/2840-01-444-7012 **Overall Length:** 4.891 inches **Overall Width:** Between 1.347 inches and 1.352 inches **End Application:** Aircraft eng tf39 **Design Type:** Axial compressor **Blade Attachment Method:** Fir tree **Criticality Code Justification: Engine Section Location:** High pressure turbine Stage Number: **Special Features:** Material external platinum aluminide coating, this is part of the ht90 mod for the tf39, original drawing name blade, high pressure turbine rotor stage 2, trail Material: Aluminum blade Shelf Life: N/a **Unit Of Measure:**

Demilitarization:

Yes - demil/mli

Fiig:

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