NSN 2840-01-447-2505

Aircraft Gas Turbine Engine Compressor Rotor - Page 1 of 1



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Overall Length:
Between 25.577 inches and 25.597 inches
End Application:
F110-129 aircraft gas turbine engine
Design Type:
Axial compressor
Criticality Code Justification:
Feat and czbj and abhp
Product Name:
Original name rotor assembly, compressor
Engine Section Location:
Compressor
Special Features:
5includes stages 1-9; hpcr stg 1-2 has improved spool oil drain holes (2) that are 1.48 time larger than the current elliptical hole; replaces
p/n 9547m14g03 and 9547m14g06; change of hole is a physical change but not functionally interchangeable
Shelf Life:
N/a
Unit Of Measure:

Demilitarization:
Yes - demil/mli
Fiig:
A10700