Aircraft Gas Turbine Engine Compressor Rotor - Page 1 of 1



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Overall Length:

Between 25.577 inches and 25.597 inches

End Application:

F110-129 aircraft gas turbine engine

Design Type:

Axial compressor

Criticality Code Justification:

Feat and czbj and abhp

Product Name:

Original name rotor assembly, compressor

Engine Section Location:

Compressor

Special Features:

5includes stages 1-9; hpcr stg 1-2 has improved spool oil drain holes (2) that are 1.48 time larger than the current elliptical hole; replaces p/n 9547m14g03 and 9547m14g06; change of hole is a physical change but not functionally interchangeable

Shelf Life:

N/a

Unit Of Measure:

Demilitarization:

Yes - demil/mli

Fiig:

A10700