NSN 2840-01-447-2868

Aircraft Gas Turbine Engine Compressor Disk And Hub - Page 1 of 1



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Overall Length:
Between 6.425 inches and 6.455 inches
Diameter:
Between 17.225 inches and 17.229 inches
End Application:
F110-ge-129 aircraft gas turbine engine
Design Type:
Axial compressor
Criticality Code Justification:
Abhp and abmz and feat
Product Name:
Original name spool, compressor rotor
Engine Section Location:
Compressor
Special Features:
T.O 2j-f110-14, fig 67, ind 5; repl p/n 1559m63g02; hpcr stg 1-2 spool oil drain holes are enlarged 1.48 larger than the current elliptical
holes to allow better drainage
Material:
Titanium alloy
Shelf Life:
N/a
Unit Of Measure:
-
Demilitarization:
Yes - demil/mli
Fiig:
A10700