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**End Application:**

F100-pw-229 aircraft gas turbine engine

**Design Type:**

Axial compressor

**Criticality Code Justification:**

Feat and czbj

**Product Name:**

Original name rotor and stator assembly, turbine rear compressor drive

**Engine Section Location:**

Compressor

**Stage Number:**

1

**Special Features:**

T.O. 2j-f100-54, fig 86, ind 1; location near the rear compressor drive turbine rotor and stator assembly, turbine stator support; use only on p/n 4084522-800, -801, -802 and 4082657-808

**Shelf Life:**

N/a

**Unit Of Measure:**

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**Demilitarization:**

Yes - demil/mli

**Fiig:**

A10700