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End Application:

F100-pw-229 aircraft gas turbine engine

Design Type:

Axial compressor

Criticality Code Justification:

Feat and czbj

Product Name:

Original name rotor and stator assembly, turbine rear compressor drive

Engine Section Location:

Compressor

Stage Number:

1

Special Features:

T.O. 2j-f100-54, fig 86, ind 1; location near the rear compressor drive turbine rotor and stator assembly, turbine stator support; use only on p/n 4084522-800, -801, -802 and 4082657-808

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

Yes - demil/mli

Fiig:

A10700