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End Application:

F100-pw-229 aircraft gas turbine engine

Criticality Code Justification:

Feat

Engine Section Location:

Combustion chamber

Special Features:

T.O. 2j-f100-54, fig 86, ind 22; located near the rear compressor drive turbine rotor and stator assembly, 1st stage turbine stator support, vanes, inner air sealing ring; use only on p/ns 4084522-800, 4082654-803 and 4082657-808

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

Yes - demil/mli

Fiig:

A530f0