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**End Application:**

F100-pw-229 aircraft gas turbine engine

**Criticality Code Justification:**

Feat

**Engine Section Location:**

Combustion chamber

**Special Features:**

T.O. 2j-f100-54, fig 86, ind 22; located near the rear compressor drive turbine rotor and stator assembly, 1st stage turbine stator support, vanes, inner air sealing ring; use only on p/ns 4084522-800, 4082654-803 and 4082657-808

**Shelf Life:**

N/a

**Unit Of Measure:**

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**Demilitarization:**

Yes - demil/mli

**Fiig:**

A530f0