NSN 2840-01-448-7033

Aircraft Gas Turbine Engine Air Seal - Page 1 of 1



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|---|
| Overall Length: |
| Between 0.154 inches and 0.166 inches |
| Overall Width: |
| Between 0.258 inches and 0.268 inches |
| Diameter: |
| Between 19.626 inches and 19.630 inches |
| End Application: |
| F100-pw-229 aircraft gas turbine engine |
| Design Type: |
| Axial compressor |
| Criticality Code Justification: |
| Abhp and abmz and abmk and czbj and feat |
| Product Name: |
| Original name brushseal, inner |
| Engine Section Location: |
| Combustion chamber |
| Stage Number: |
| 1 |
| Special Features: |
| T.O. 2j-f100-54, fig 86, ind 19g; seal contains brush wire pack that has 2250-2650 wires per circumferential inch; use only on p/ns |
| 4083194, 4083560 and 4083565; located near the rear compressor drive turbine rotor and stator assembly, turbine stator support |
| Material: |
| Nickel alloy |
| Material Specification: |
| Ams5599 assn standard 1st material response or ams5666 assn standard 2nd material response |
| Shelf Life: |
| N/a |
| Unit Of Measure: |
| |
| Demilitarization: |
| Yes - demil/mli |
| Fiig: |
| A10700 |