

View Online at <https://aerobasegroup.com/nsn/2840-01-448-7035>

Overall Length:

Between 0.258 inches and 0.268 inches

Diameter:

Between 25.004 inches and 25.008 inches

End Application:

F100-pw-229 aircraft gas turbine engine

Design Type:

Axial compressor

Criticality Code Justification:

Czbj and abhp and abmz and feat

Product Name:

Original name seal brush, combustion chamber, outer assembly

Engine Section Location:

Combustion chamber

Stage Number:

1

Special Features:

T.O. 2j-f100-54, fig 86, ind 19f; use only on p/ns 4083194, 4083560, and 4083565; contains wire brush, 2250-2650 wires per circumferential inch; located near the rear compressor drive turbine rotor and stator assembly; turbine stator support and combustion chamber

Material:

Nickel alloy

Material Specification:

Ams5599 assn standard 1st material response or ams5666 assn standard 2nd material response

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

Yes - demil/mli

Fiig:

A10700