## NSN 2840-01-459-1286

Aircraft Gas Turbine Engine Compressor Disk And Hub - Page 1 of 1



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Overall Length:
Between 6.425 inches and 6.455 inches
Diameter:
Between 17.225 inches and 17.229 inches
End Application:
F118-ge-100 aircraft gas turbine engine
Design Type:
Axial compressor
Criticality Code Justification:
Abhp and abmz and feat
Product Name:
Original name hpc compressor spool
Engine Section Location:
Compressor
Special Features:
Pt.O. 2j-f110-4, fig 67, ind 5; hpcr stg 1-2, spool oil drain holes are enlarged 1.48 larger than the current elliptical holes to allow better
drainage; can only be used on the f118-ge-100 aircraft jet turbine engine
Material:
Titanium alloy
Shelf Life:
N/a
Unit Of Measure:
<del></del>
Demilitarization:
Yes - demil/mli
Fiig:
A10700