

View Online at <https://aerobasegroup.com/nsn/5340-00-344-2956>

Fabrication Method (non-core):

Machined

End Application:

Case assembly, turbine; w/s: engine, aircraft tf33-pw-102 (c-135e, ec-135h/k/p); engine, aircraft, tf-33-p-7 (c-141a/b); aircraft, stratofortress b-52; engine, aircraft tf33-p-3/5/9 (c/ec-135, b-52h)

Second Leg Width:

1.509 inches

Second Leg Corner Radius:

0.250 inches

First Leg Hole Diameter:

0.224 inches

Second Leg Hole Diameter:

0.286 inches

Second Leg Distance Between Hole Centers Along Width:

1.009 inches

Second Leg Distance From Edge To Hole Center Along Width:

0.250 inches

Second Leg Distance From Edge To Hole Center Along Length:

0.250 inches

First Leg Thickness:

0.062 inches

Second Leg Thickness:

0.062 inches

First Leg Width:

0.510 inches

First Leg Distance From Edge To Hole Center Along Width:

0.250 inches

First Leg Distance From Edge To Hole Center Along Length:

0.250 inches

First Leg End Radius:

0.375 inches

First Leg Style:

End radius

First Leg Hole Arrangement Style:

One hole

Second Leg Style:

Corner radius

Second Leg Hole Arrangement Style:

Two holes

Second Leg Relationship To First Leg:

Not identical

First Leg Angle:

130.0 degrees

Second Leg Offset Angle:

80.9 degrees

Material:

Steel comp 410

Material Specification:

66 federal standard single material response

Style Designator:

Offset angle

Fsc Application Data:

Engine, jet, aircraft

Shelf Life:

N/a

Unit Of Measure:

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Demilitarization:

No

Fig:

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